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MODEL BASED DESIGN

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References

- Slides are from Edward A. Lee & Sanjit Seshia, UC Berkeley, EECS 149 Fall 2013
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- Some slides from Philip Asare w/ Joanne Dugan and Ron Williams from UVA

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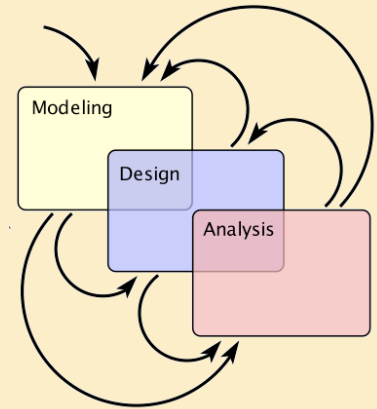
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Modeling, Design, Analysis

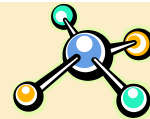
- **Modeling** is the process of gaining a deeper understanding of a system through imitation. Models specify **what** a system does.
- **Design** is the structured creation of artifacts. It specifies **how** a system does what it does.
- **Analysis** is the process of gaining a deeper understanding of a system through dissection. It specifies **why** a system does what it does (or fails to do what a model says it should do).



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What is Modelling?



- Developing insight about a system, process, or artefact through imitation.
- A *model* is the artefact that imitates the system, process, or artefact of interest.
- A *mathematical model* is in the form of a set of definitions and mathematical formulas/objects.

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The Kopetz Principle



Prof. Dr. Hermann Kopetz

■ Many (predictive) properties that we assert about systems (determinism, timeliness, reliability, safety) are in fact not properties of an *implemented* system, but rather properties of a *model* of the system.

■ We can make definitive statements about *models*, from which we can *infer* properties of system realizations. The validity of this inference depends on *model fidelity*, which is always approximate.

(paraphrased)

What is Model-Based Design?

1. Create a mathematical model of all the parts of the embedded system

- *Physical world*
- *Control system*
- *Software environment*
- *Hardware platform*
- *Network*
- *Sensors and actuators*

Different sub-systems,
different approaches to
Modelling

2. Construct the implementation from the model

- *Goal: automate this construction, like a compiler*
- *In practice, only portions are automatically constructed*

Modelling Techniques

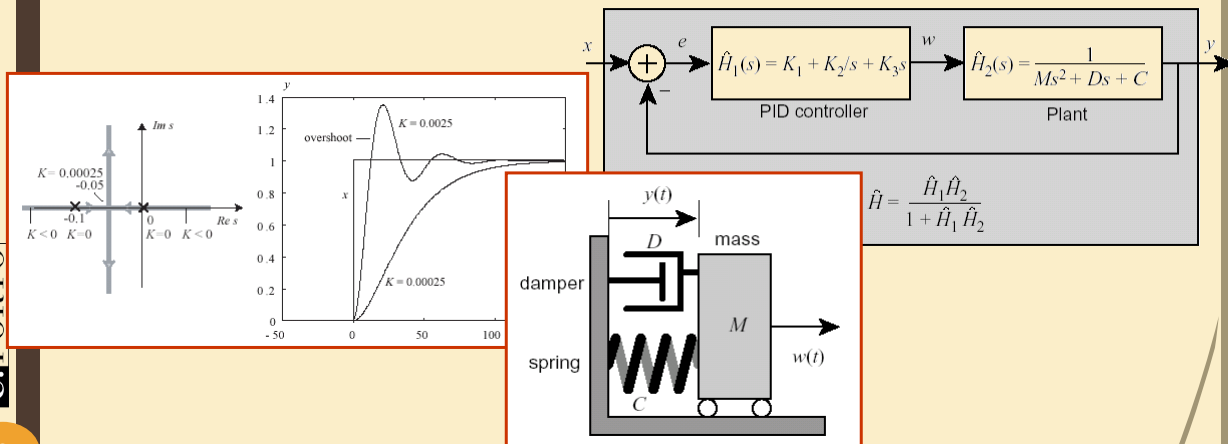
- Models that are abstractions of system dynamics
 - (how things change over time)
- Examples:
 - Modelling physical phenomena – Ordinary Differential Equations (ODEs)
 - Feedback control systems – time-domain Modelling
 - Modelling modal behaviour – FSMs, hybrid automata
 - Modelling sensors and actuators – calibration, noise
 - Modelling software – concurrency, real-time models
 - Modelling networks – latencies, error rates, packet loss

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Modelling of Continuous Dynamics

- Ordinary differential equations, Laplace transforms, feedback control systems, stability analysis, robustness analysis, ...



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An Example: Modelling Helicopter Dynamics

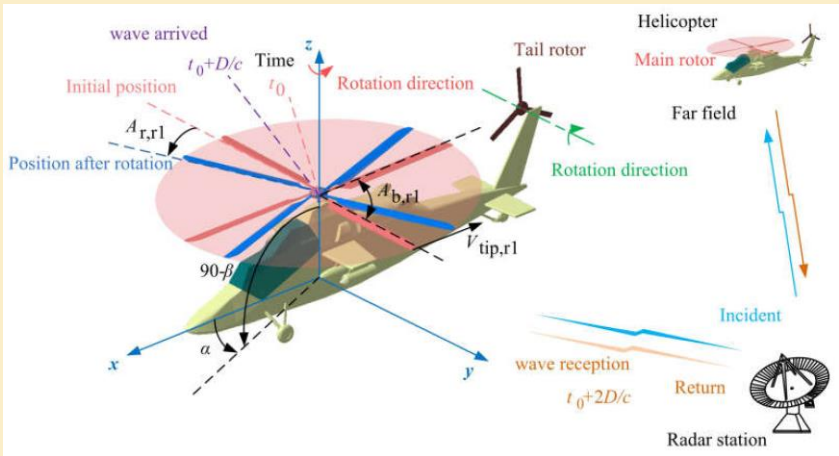
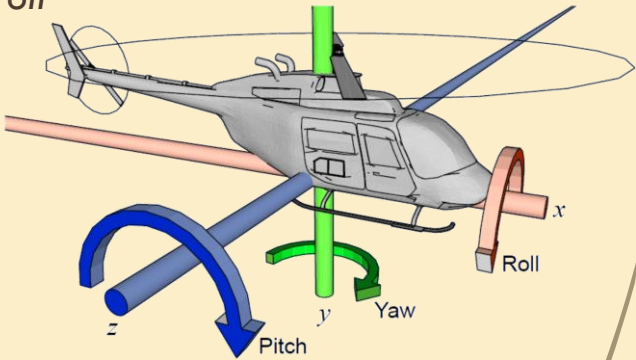


Image from Zhou Z, Huang J. [Influence of Rotor Dynamic Scattering on Helicopter Radar Cross-Section](#). *Sensors*. 2020; 20(7):2097.

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Modelling Physical Motion

- Six degrees of freedom:
 - Position: x, y, z
 - Orientation: pitch, yaw, roll



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Notation

- Position is given by three functions:

$$x: \mathbb{R} \rightarrow \mathbb{R}$$

$$y: \mathbb{R} \rightarrow \mathbb{R}$$

$$z: \mathbb{R} \rightarrow \mathbb{R}$$

- Where the domain \mathbb{R} represents time and the co-domain \mathbb{R} (range) represents position along the axis.

- Collecting into a vector (x):

$$x: \mathbb{R} \rightarrow \mathbb{R}^3$$

- Position at time $t \in \mathbb{R}$ is $x(t) \in \mathbb{R}^3$

Notation

- Velocity:

$$\dot{x}: \mathbb{R} \rightarrow \mathbb{R}^3$$

- Is the derivative, $\forall t \in \mathbb{R}$,

$$\dot{x}(t) = \frac{d}{dt} x(t)$$

- Acceleration $\ddot{x}: \mathbb{R} \rightarrow \mathbb{R}^3$ is the second derivative

$$\ddot{x}(t) = \frac{d^2}{dt^2} x(t)$$

Newton's Second Law

- Force on an object is $F: \mathbb{R} \rightarrow \mathbb{R}^3$
- Newton's second law states $\forall t \in \mathbb{R}$,

$$F(t) = M\ddot{x}(t)$$
- where M is the mass.

- Converting to integral equation:

$$\begin{aligned} x(t) &= x(0) + \int_0^t \dot{x}(\tau) d\tau \\ &= x(0) + \int_0^t [\dot{x}(0) + \int_0^\tau \ddot{x}(\alpha) d\alpha] d\tau \\ &= x(0) + t\dot{x}(0) + \frac{1}{M} \int_0^t \int_0^\tau F(\alpha) d\alpha d\tau \end{aligned}$$

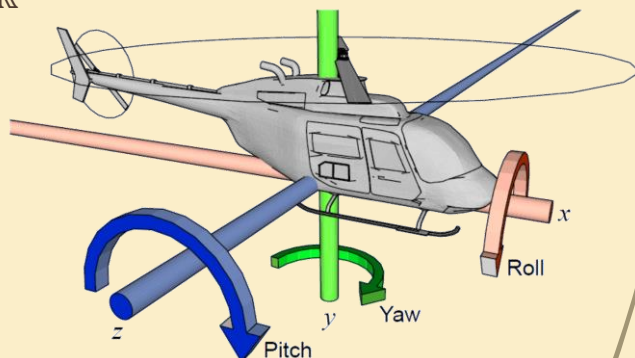
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Orientation

- Orientation: $\theta: \mathbb{R} \rightarrow \mathbb{R}^3$
- Angular velocity: $\dot{\theta}: \mathbb{R} \rightarrow \mathbb{R}^3$
- Angular acceleration: $\ddot{\theta}: \mathbb{R} \rightarrow \mathbb{R}^3$
- Torque: $T: \mathbb{R} \rightarrow \mathbb{R}^3$

$$\theta(t) = \begin{bmatrix} \theta_x(t) \\ \theta_y(t) \\ \theta_z(t) \end{bmatrix} = \begin{bmatrix} \text{roll} \\ \text{yaw} \\ \text{pitch} \end{bmatrix}$$



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Point mass rotating around a fixed axis

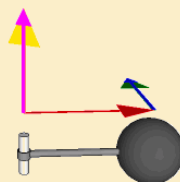
- Angular version of force is torque.
- Just as force is a push or a pull, a torque is a twist.
- Units: newton-meters/radian, Joules/radian
- Note that radians are meter/meter (2π meters of circumference per 1 meter of radius), so as units, are optional.
- Radius of arm: $r \in \mathbb{R}$
- Force orthogonal to arm: $f \in \mathbb{R}$
- Mass of object: $m \in \mathbb{R}$

$$T_y(t) = r f(t)$$

angular momentum, momentum

$$\tau = r \times F$$

$$L = r \times p$$



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Rotational Version of Newton's Second Law

$$T(t) = \frac{d}{dt} (I(t) \dot{\theta}(t))$$

- Where $I(t)$ is a 3×3 matrix called the moment of inertia tensor:

$$\begin{bmatrix} T_x(t) \\ T_y(t) \\ T_z(t) \end{bmatrix} = \frac{d}{dt} \left(\begin{bmatrix} I_{xx}(t) & I_{xy}(t) & I_{xz}(t) \\ I_{yx}(t) & I_{yy}(t) & I_{yz}(t) \\ I_{zx}(t) & I_{zy}(t) & I_{zz}(t) \end{bmatrix} \begin{bmatrix} \dot{\theta}_x(t) \\ \dot{\theta}_y(t) \\ \dot{\theta}_z(t) \end{bmatrix} \right)$$

- Here, for example, $T_y(t)$ is the net torque around the y axis (which would cause changes in yaw), $I_{yx}(t)$ is the inertia that determines how acceleration around the x axis is related to torque around the y axis.

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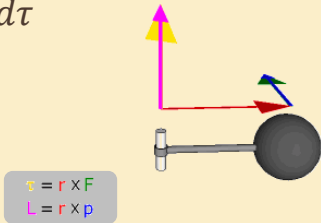
Simple Example

- Yaw Dynamics

$$T_y(t) = I_{yy}\ddot{\theta}_y(t)$$

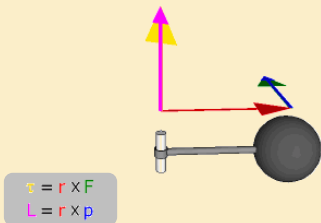
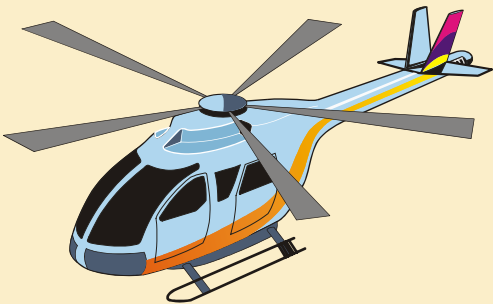
- To account for initial angular velocity:

$$\dot{\theta}_y(t) = \dot{\theta}_y(0) + \frac{1}{I_{yy}} \int_0^t T_y(\tau) d\tau$$



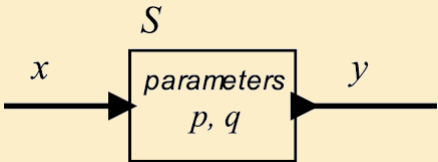
Feedback Control Problem

- A helicopter without a tail rotor, like the one below, will spin uncontrollably due to the torque induced by friction in the rotor shaft.
- Control system problem: Apply torque using the tail rotor to counterbalance the torque of the top rotor.



Actor Model of Systems

- A system is a function that accepts an input signal and yields an output signal.



- The domain and range of the system function are sets of signals, which themselves are functions.

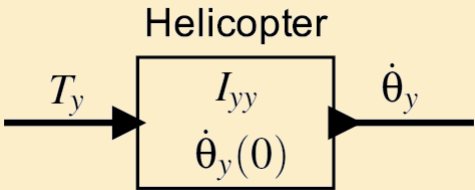
$$x: \mathbb{R} \rightarrow \mathbb{R}, \quad y: \mathbb{R} \rightarrow \mathbb{R}$$

- Parameters may affect the definition of the function S .

$$S: X \rightarrow Y$$
$$X = Y = \mathbb{R} \rightarrow \mathbb{R}$$

Actor model of the helicopter

- Input is the net torque of the tail rotor and the top rotor. Output is the angular velocity around the y axis.



Parameters of the model are shown in the box. The input and output relation is given by the equation.

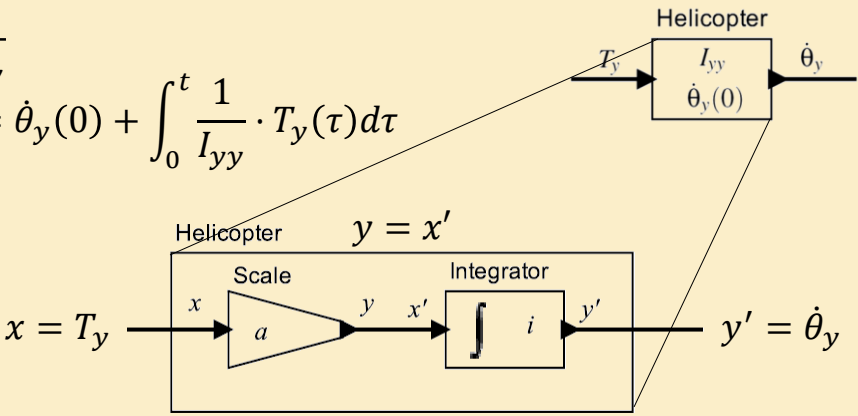
$$\dot{\theta}_y(t) = \dot{\theta}_y(0) + \frac{1}{I_{yy}} \int_0^t T_y(\tau) d\tau$$

Composition of actor models

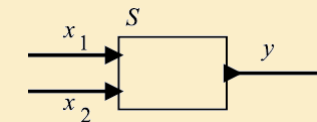
$i = \dot{\theta}_y(0)$ (initial value of integration)

$a = \frac{1}{I_{yy}}$

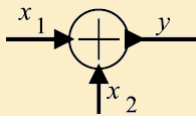
$\dot{\theta}_y(t) = \dot{\theta}_y(0) + \int_0^t \frac{1}{I_{yy}} \cdot T_y(\tau) d\tau$



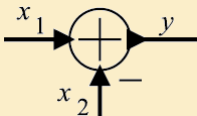
Actor models with multiple inputs



$S: (\mathbb{R} \rightarrow \mathbb{R})^2 \rightarrow (\mathbb{R} \rightarrow \mathbb{R})$



$\forall t \in \mathbb{R}, \quad y(t) = x_1(t) + x_2(t) \quad (S(x_1, x_2))(t) = x_1(t) - x_2(t)$



Stability: Bounded-Input Bounded-Output

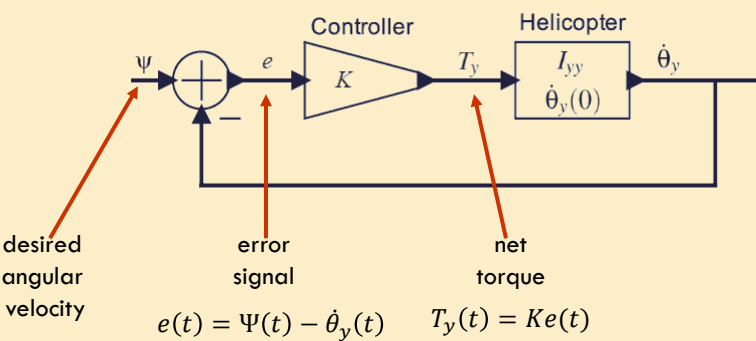
- A system is bounded-input bounded-output (BIBO) stable if for an input that is bounded for all time, the output remains bounded for all time

- More formally

Let $x(t)$ be the input signal and $y(t)$ be the output signal.
The input is bounded if there is some real number $A < \infty$ such that $x(t) < A, \forall t \in \mathbb{R}$ and similarly the output is bounded if there is some real number B such that $y(t) < B, \forall t \in \mathbb{R}$. The system is stable (in the BIBO sense) if for a bounded input $x(t)$ there exists some bound B for the output $y(t)$.

The helicopter is not BIBO stable. Why?

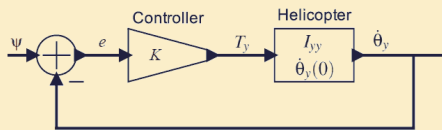
Stability: Proportional controller



$$\begin{aligned} \dot{\theta}_y(t) &= \dot{\theta}_y(0) + \frac{1}{I_{yy}} \int_0^t T_y(\tau) d\tau \\ &= \dot{\theta}_y(0) + \frac{K}{I_{yy}} \int_0^t (\psi(\tau) - \dot{\theta}_y(\tau)) d\tau \end{aligned}$$

Note that the angular velocity appears on both sides, so this equation is not trivial to solve.

Behaviour of the controller



$$\dot{\theta}_y(t) = \dot{\theta}_y(0) + \frac{K}{I_{yy}} \int_0^t \left(\Psi(\tau) - \dot{\theta}_y(\tau) \right) d\tau$$

Desired angular velocity: $\Psi(\tau) = 0$

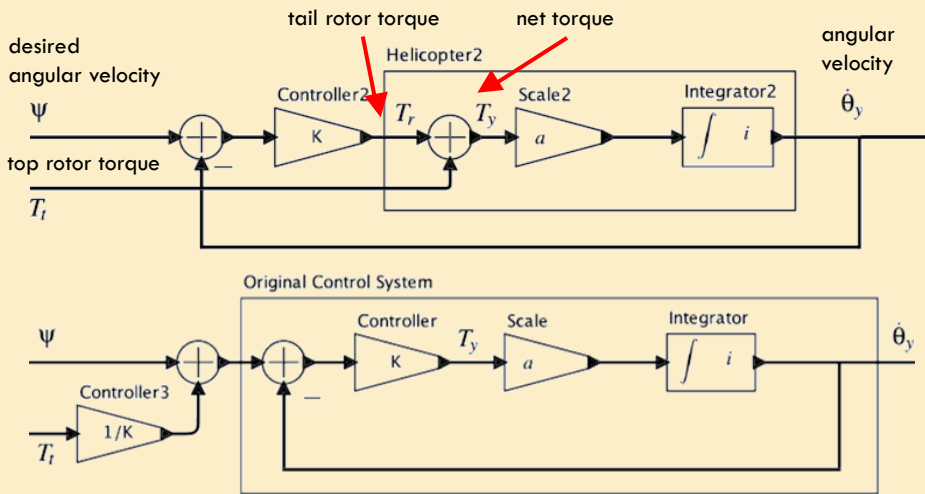
Simplifies differential equation to:

$$\dot{\theta}_y(t) = \dot{\theta}_y(0) - \frac{K}{I_{yy}} \int_0^t \dot{\theta}_y(\tau) d\tau$$

Which can be solved as follows (see textbook):

$$\dot{\theta}_y(t) = \dot{\theta}_y(0) e^{-Kt/I_{yy}} u(t)$$

More Realistic Scenario



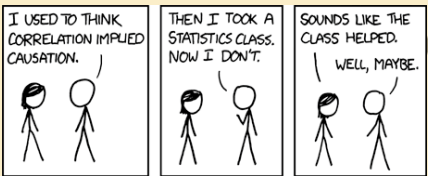
System Properties – Causality

- The **current output** can only depend on **the current and past inputs**
- More formally:

Let $x|_{t \leq \tau}$ be an input that has values only at times $t \leq \tau$
(this is called a restriction in time)

IF $x_1|_{t \leq \tau} = x_2|_{t \leq \tau}$
THEN a system S for which x_1 and x_2 are valid inputs
is causal if and only if $S(x_1)|_{t \leq \tau} = S(x_2)|_{t \leq \tau}$

Source: XKCD -Correlation



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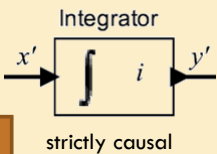
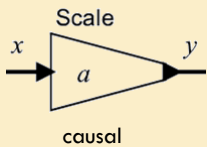
System Properties – Strict Causality

Strict causality \Rightarrow causality

- The **current output** can only depend on **the past inputs**
- More formally:

Let $x|_{t < \tau}$ be an input that has values only at times $t < \tau$

IF $x_1|_{t < \tau} = x_2|_{t < \tau}$
THEN a system S for which x_1 and x_2 are valid inputs
is strictly causal if and only if $S(x_1)|_{t \leq \tau} = S(x_2)|_{t \leq \tau}$
Note that the output of S is for times up to **and including** t



For the integrator, the value of x at time τ is irrelevant for $y(\tau)$. So, x_1 can differ from x_2 at time τ .

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System Properties – Memoryless

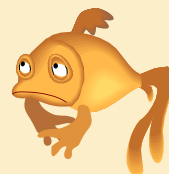
- The **current output** only depends on the **current input**
- More formally:

Remember that if x is a signal, then x is a function,
hence $x(t)$ is the value at time t of this function

Additionally, the system S is also a function and
hence $(S(x))(t)$ is the output of the system at t for an input signal x

A system S is memoryless if there is some function
 $f: A \rightarrow B$ such that $(S(x))(t) = f(x(t))$

(i.e., the output at t depends on the input at t only)



Note however that Fish's memories last for months, say scientists, the Guardian

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System Properties – Linearity

- Must satisfy superposition

- Additivity

$$S(x_1 + x_2) = S(x_1) + S(x_2)$$

- Homogeneity

$$S(a \cdot x) = a \cdot S(x)$$

- Superposition

$$S(a \cdot x_1 + b \cdot x_2) = a \cdot S(x_1) + b \cdot S(x_2)$$



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Is the helicopter system linear?

$$\dot{\theta}_y(t) = \dot{\theta}_y(0) + \int_0^t \frac{1}{I_{yy}} \cdot T_y(\tau) d\tau$$

$$S(T_y) = \dot{\theta}_y(0) + \int_0^t \frac{1}{I_{yy}} \cdot T_y(\tau) d\tau$$

$$S(a \cdot T_y) = \dot{\theta}_y(0) + \int_0^t \frac{1}{I_{yy}} \cdot a \cdot T_y(\tau) d\tau$$

$$a \cdot S(T_y) = a \cdot \left[\dot{\theta}_y(0) + \int_0^t \frac{1}{I_{yy}} \cdot T_y(\tau) d\tau \right]$$

$$S(a \cdot T_y) \neq a \cdot S(T_y)$$

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System Properties – Time-Invariance



- The output of the system acting on a delayed version of the input is equal to the delayed version of the output of the system acting on the original system.
- More formally

Let D_τ be an actor (system) that delays a signal
such that $D_\tau(x(t)) = x(t - \tau)$

Then a system S is time invariant if and only if $S(D_\tau(x)) = D_\tau(S(x))$

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System Properties

■ Time Invariance Example

- Let $x(t) = \sin(t)$ and $S(x) = a \cdot x$
- Then $S(x(t - \tau)) = S(\sin(t - \tau)) = a \cdot \sin(t - \tau)$
- and $(S(x))(t - \tau) = S(\sin(t))|_{t=t-\tau} = a \cdot \sin(t - \tau)$
- $S(x(t - \tau)) = (S(x))(t - \tau)$

■ Time Variance Example

- Let $x(t) = \sin(t)$ and $S(x) = t \cdot x$
- Then $S(x(t - \tau)) = S(\sin(t - \tau)) = t \cdot \sin(t - \tau)$
- and $(S(x))(t - \tau) =$

$$S(\sin(t))|_{t=t-\tau} = (t - \tau) \cdot \sin(t - \tau)$$
- $S(x(t - \tau)) \neq (S(x))(t - \tau)$

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Key Concepts

- Models describe physical dynamics.
 - Specifications are executable models.
 - Models are composed to form designs.
 - Models evolve during design.
 - Deployed code may be (partially) generated from models.
 - Modelling languages have semantics.
 - Modelling languages themselves may be modelled (meta models)
- For embedded systems, this is about
 - Time
 - Concurrency
 - Dynamics

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Summary

- Signals and systems: formal definition
- Continuous dynamical systems with Newton mechanics
- Actor models
- Some properties of systems